## EFI-650 ELECTRONIC FLIGHT INSTRUMENT

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Regional and business jet operators are reducing operating costs and improving the flight experience with the EFI-650 display. This plug-and-play replacement addresses the aging CRT obsolescence, while offering huge gains in reliability, reduces weight, and lowers power consumption over legacy systems.

The EFI-650's mean-time between failures (MTBF) is greater than 25,000 hours which is a 50% improvement over CRTs. This reliability reduces operational disruptions and the risk of AOGs, as well as the need to provision and hold more spares. The LCD display is 50% lighter than standard CRTs, therefore reducing operational costs and offering exceptional clarity and brightness to the pilots.

- Drop-in replacement for the Honeywell ED-800 CRT display
- No change to cockpit panel or LRU mounting prov diminates Cathode Ray Tube

(CRT) and high-voltage power supply for improved reliability

- > Reduces life-cycle costs
- > Reduces display weight by 50%
- > Provides growth provision to

EF1-650			
OPTICAL PERFORMANCE			
T'I a a	3.4" (H) x 4.53" (W)		
rr a ii.	6-bits per color		
. É 1 m	640 x 480 (VGA)		
	262,144 possible; Video interface supports palette of 13 colors + black		
l'La	0.1 fL to ~150 fL		
T ) ). ,, r	±45° horizontal; +45°, -23° vertical		
'i <sub>y</sub> , l ⊸, a, a <u>'</u> i	>3:1 across viewing envelope		
PHYSICAL DESCRIPTION			
νι, · · · · · · · · · · · · · · · · · ·	w/inclinometer: 5.1" x 6.1" x 11.0" w/out inclinometer: 5.1" x 6.1" x 10.5"		
۹ <sub>γ</sub> .	Approx. 5 lb.		
as hy	Passive convective		
a' <b>ı.</b> ,	Integrated heater		
il a, 🛥	28VDC per DO-160G, Sec. 16, Cat. B; Operate thru 200 ms power interrupt <20 Watts		
	<35 Watts w/heater on		
/ al	Photosensors for automatic dimming in response to ambient light		
FUNCTIONAL DESCRIPTION			
Tì 💶 ar	Analog (stroke and raster) video		
T` <b>\</b> ⊸ <sub>1</sub>	Analog (stroke and raster)		
'ı <sub>y</sub> -aa <sub>r</sub>	Analog and discrete		
. () 'ı. 'ı ()	Inclinometer is dimmable under control of aircraft lighting bus		

APPROVED STCs	
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EFI-650		
ENVIRONMENTAL PERFOR		1606)
$a_1 \otimes b_1$	INANCE (DO-	Sec. 4, Cat. A2
a`ı. <sub>y</sub> , l.		Sec. 4.5.2, -20° C
	1 -	Sec. 4.5.1, -40° C
$\mathbf{h}_{\mathbf{y}}, \mathbf{h}_{\mathbf{y}} = \mathbf{h}_{\mathbf{y}} \mathbf{h}_{\mathbf{y}}$		Sec. 4.5.3, +70° C
$\mathbf{h}_{\mathbf{y}},  \mathbf{h}_{\mathbf{y}} = \mathbf{a} \mathbf{h}_{\mathbf{y}},  \mathbf{h}_{\mathbf{y}}$		Sec. 4.5.4, +70° C
	( _)[] (	)Be(c.)4[6.5], 24 <u>0</u> n)in @(a40°,C) [
	((	Sec. 4.5.1, -55° C

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